# Goodbye TLES, held OMMs!

Juan Luis Cano Rodríguez <hello@juanlu.space> 2020-12-13 @ OSCW 2020



## Overview

- 1. What is a TLE?
- 2. What problems do they have?
- 3. What is the new proposed format, OMM?
- 4. Some details to consider
- 5. OMM support in the open source ecosystem
- 6. Conclusions and Q&A

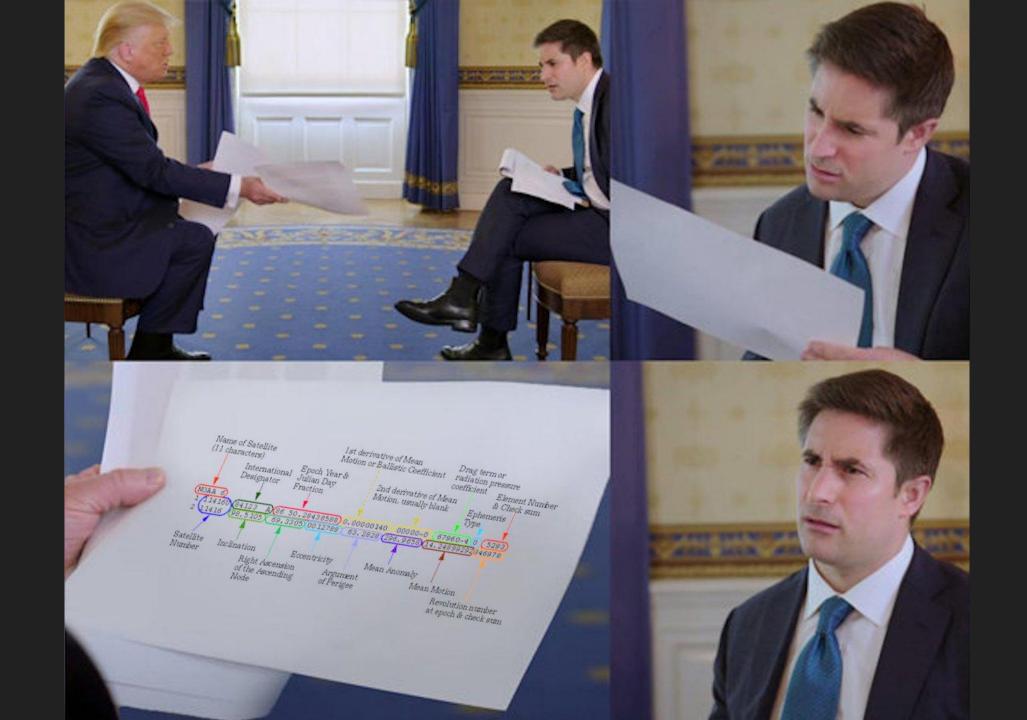
# **1. What is a TLE?** Well...

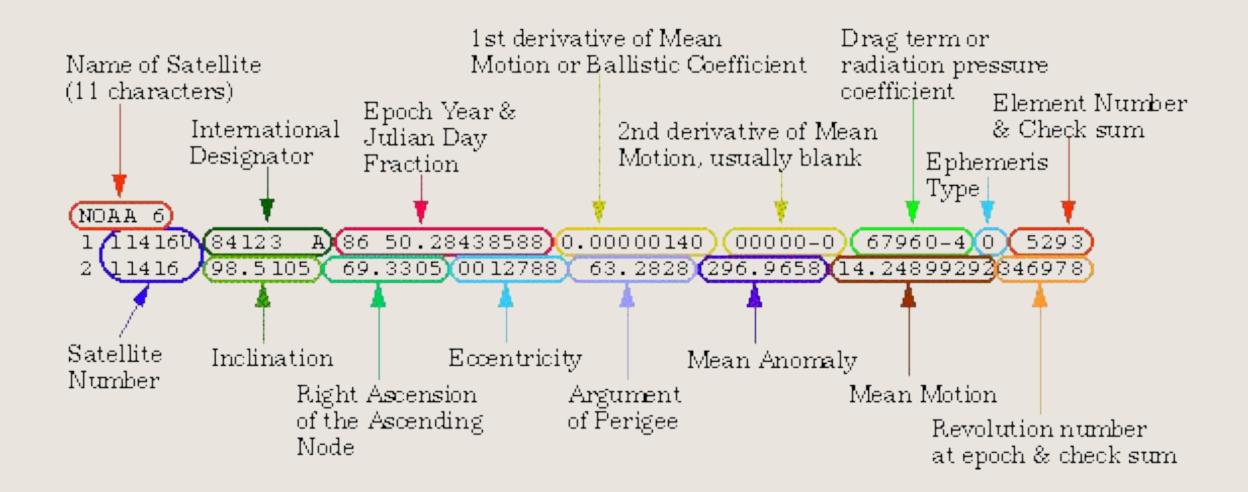


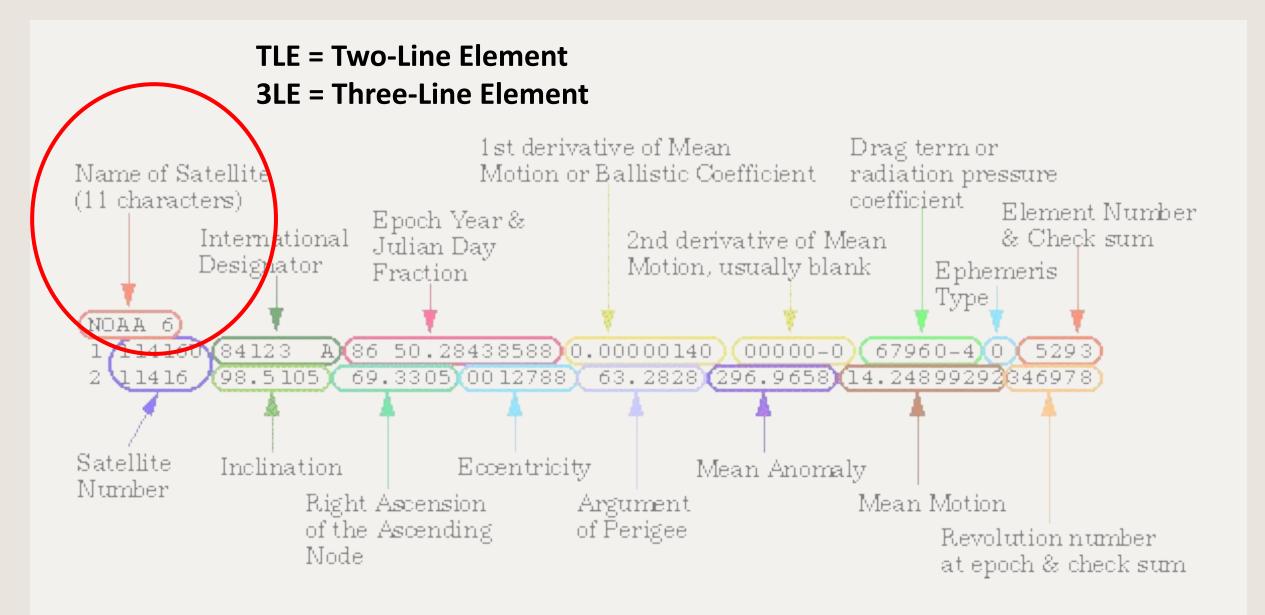
How about "a zombie data encoding system dating back to the punched card days with an amazing Y2056 problem (among other issues)"?

10:26 AM · Sep 24, 2020

See pleiszenburg.de's other Tweets

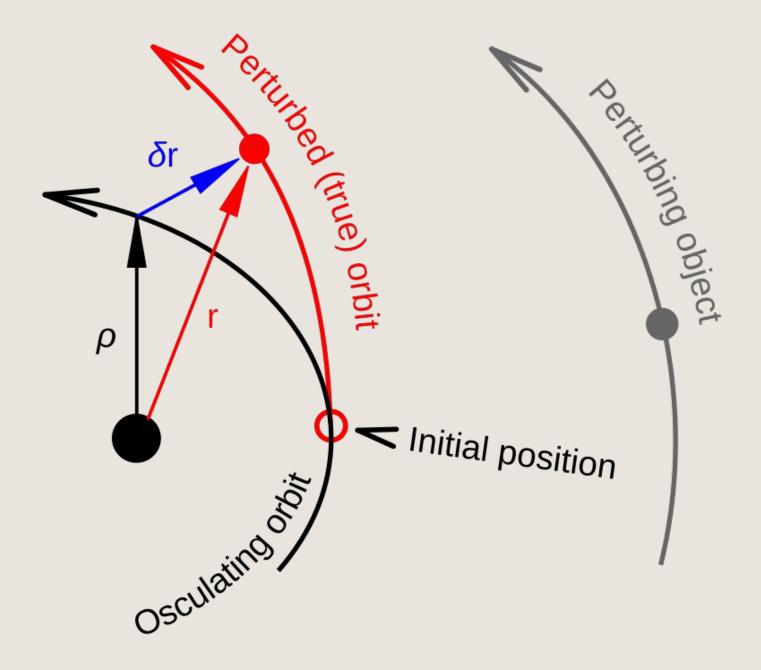


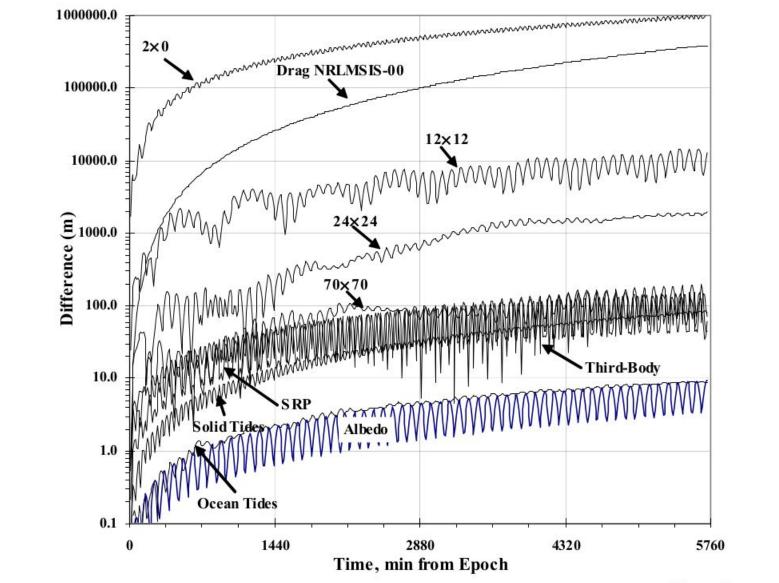




"The US government has provided GP or *general perturbations orbital data* to the rest of the world since the 1970s. These data are produced by fitting observations from the US Space Surveillance Network (SSN) to produce Brouwer mean elements using the SGP4 or Simplified General Perturbations 4 orbit propagator.

Many of you are familiar with this *data in the form of TLEs or Two-Line Element Sets*."





**Figure 9-17.** Force Model Comparisons — LEO 500 × 500 km, 97.6°. This figure shows RSS position differences for several forces. This is the JERS satellite, NORAD# 21867 which is also Sun-synchronous.

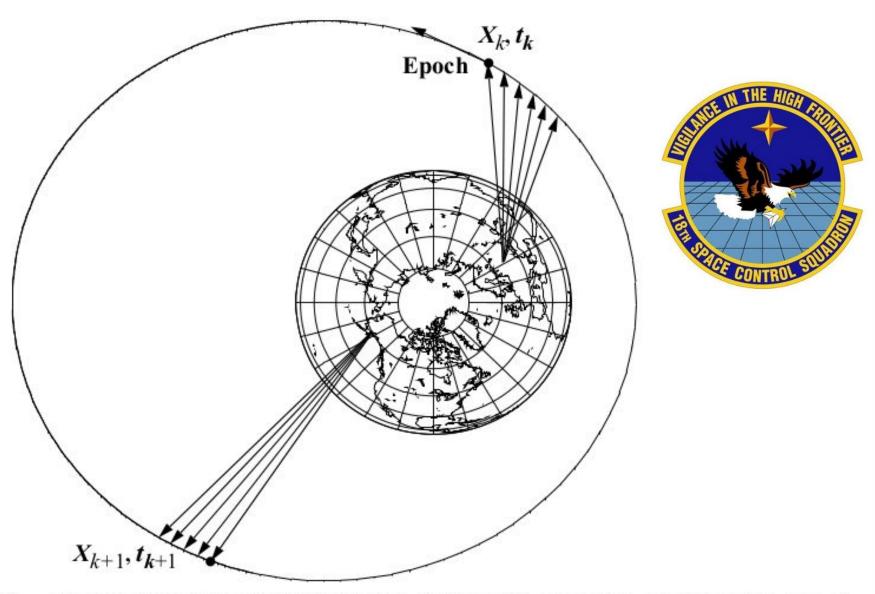


Figure 10-7. Observations from Multiple Passes. Differential correction minimizes the sum of the squares of the residuals by referring all corrections to a single state at a *fixed epoch*,  $t_k$ . The Kalman filter finds a state update at each observation time,  $t_{k+1}$ .

## NORAD Two-Line Element Sets Current Data

Today from The Center for Space Standards & Innovation

## Current as of 2020 Sep 25 14:06:01 UTC (Day 269)

### System Notices

Future Availability of TLE Data Last Updated 2007 May 16

### Supplemental TLE Data

### Space Track TLE Retriever 3

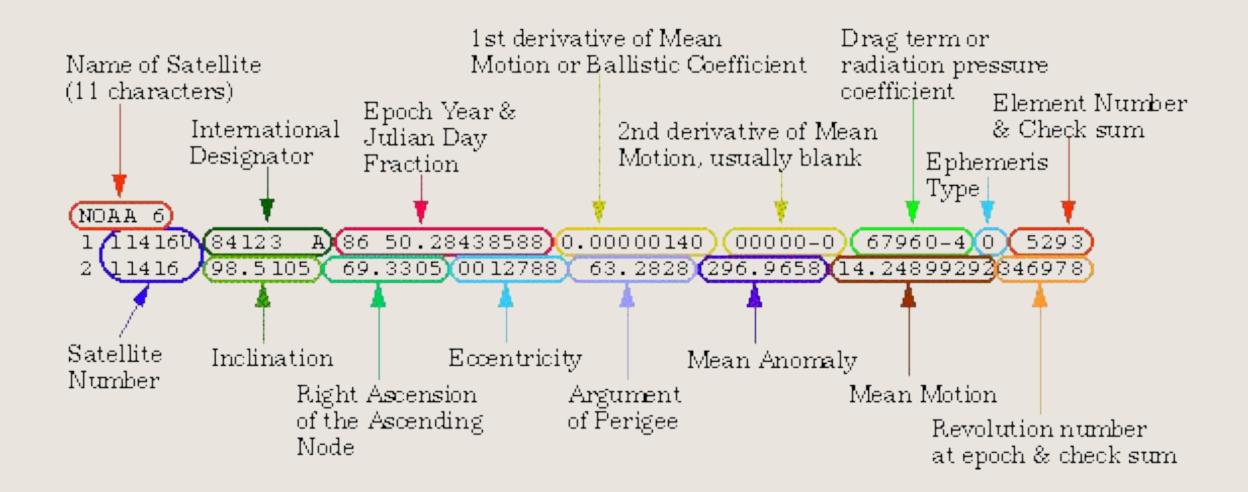
## Space Track Data Access

4	Special-Interest Satellites				
Last 30 Days' Launches 🎛 🕀					
Space Stations 🎛 🕀					
100 (or so) Brightest 🖽 🕀					
Active Satellites 🎛 🕀					
Analyst Satellites 🎛 🕀					
Indian ASAT Test Debris 🎛 🕀					
FENGYUN 1C Debris 🎛 🕀					
IRIDIUM 33 Debris 🆽 🕀					
COSMOS 2251 Debris 🎛 🕀					
Weather & Earth Resources Satellites					
	Weather 🌐 🕀				
NOAA 🎛 🕀	GOES 🎛 🕀				
	Earth Resources 🎛 🕀				

# **2. What problems do TLEs have?** Quite a few, to be honest

All highly accurate propagation techniques transmit elements using as many digits as possible to minimize unnecessary errors—we want to transmit with the same precision we use to determine the element set. Unfortunately, formats like the TLE set introduce uncertainty by arbitrarily limiting the number of decimal digits for each parameter. Because the element set has finite digits for each variable, we can determine the maximum accuracy available at the epoch through the transmission format. Recognize that, with only eight decimal places given in the epoch time value, it is accurate only to about 0.0004 seconds. We determine this accuracy by assuming the original value is unknown to  $\pm 5 \times 10^{-9}$  days, or about  $\pm 4.3 \times 10^{-4}$  seconds.

Now consider a roughly circular orbit with an altitude of 390 km and a velocity of about 7.6 km/s. The satellite travels about 4 m in 0.000  $43^{s}$ . The eccentricity is given to seven decimal places. This introduces an error at epoch which for a GEO satellite is about 2 m ( $r \approx a\Delta e$ ). The angles provide only four decimal places, which introduce an uncertainty at epoch of about 6 m for LEO satellites and about 35 m for GEO satellites. Remember that these estimates are based solely on the given data and have nothing to do with their mathematical formation and subsequent use. For precise orbit determination, even these errors probably aren't acceptable.





## brianweeden 🤣 @brianweeden · Dec 3, 2020

## Replying to @brianweeden

y

5/ Hence why the new S-Band Space Fence is a big deal. For the first time, we can track objects down to a few cm (lower limit has not been publicly revealed) well enough to get orbits lockheedmartin.com/en-us/products...





6/ However, the volume of data output by the Space Fence is more than the existing computer systems were designed to handle, and the increase in objects will far exceed the 69,999 object limit in the legacy catalog, as noted by the GAO (gao.gov/assets/660/653...)

## Space-Track @SpaceTrackOrg · Nov 25, 2020

The #satellite catalog is growing faster than ever. When @18SPCS begins to publish Space Fence elsets for #SpaceDebris object numbers above 270,000, we will be ready. Users will be able to access them through our GP and GP\_History API classes: space-track.org/documentation#...



## Space-Track

@SpaceTrackOrg

For systems that can't transition directly to the flexible CCSDS OMM format for the full range of 9-digit object numbers, @SpaceForceDoD developed #Alpha5 numbering schema as a stopgap that allows their sensors to use up to 339,999 objects in the legacy fixed-width TLE/3LE format 11:50 PM · Nov 25, 2020

# 3. What is the new proposed format, OMM? <sup>You're gonna love it!</sup>





## A New Way to Obtain GP Data (aka TLEs)

by Dr. T.S. Kelso 2020 May 27

### Background

The US government has provided GP or *general perturbations* orbital data to the rest of the world since the 1970s. These data are produced by fitting observations from the US Space Surveillance Network (SSN) to produce Brouwer mean elements using the SGP4 or *Simplified General Perturbations 4* orbit propagator.



**Recommendation for Space Data System Standards** 

## ORBIT DATA MESSAGES

**RECOMMENDED STANDARD** 

CCSDS 502.0-B-2

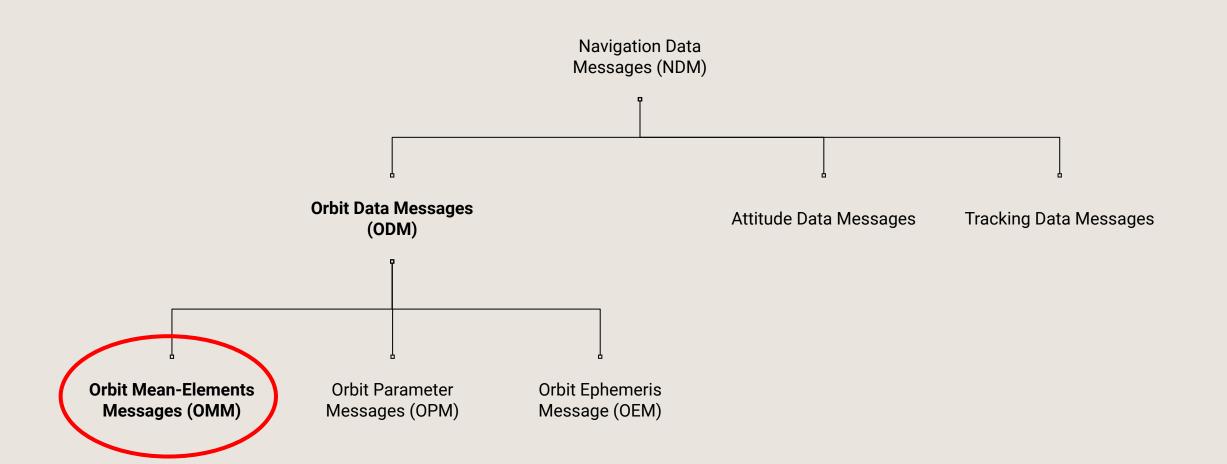
Note: This current issue includes all updates through Technical Corrigendum 1, dated May 2012.

BLUE BOOK November 2009



OBJECT_NAME:	"STARLINK-1329"
OBJECT_ID:	"2020-025A"
EPOCH:	"2020-09-24T22:00:01.999584"
MEAN_MOTION:	15.05572894
ECCENTRICITY:	0.0001294
INCLINATION:	53.0005
RA_OF_ASC_NODE:	48.0334
ARG_OF_PERICENTER:	99.318
MEAN_ANOMALY:	18.6926
EPHEMERIS_TYPE:	Θ
CLASSIFICATION_TYPE:	"U"
NORAD_CAT_ID:	45531
ELEMENT_SET_NO:	999
REV_AT_EPOCH:	177
BSTAR:	0.00097839
MEAN_MOTION_DOT:	0.00014001
MEAN_MOTION_DDOT:	Θ





#### CO

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"OBJECT NAME": "NUSAT-8 (MARIE)", "OBJECT ID": "2020-003C", "EPOCH": "2020-12-13T01:18:50.258304", "MEAN MOTION": 15.27921509, "ECCENTRICITY": 0.0012107, "INCLINATION": 97.2988, "RA OF ASC NODE": 50.6425, "ARG OF PERICENTER": 162.6718, "MEAN ANOMALY": 281.7748, "EPHEMERIS TYPE": 0, "CLASSIFICATION TYPE": "U", "NORAD CAT ID": 45018, "ELEMENT SET NO": 999, "REV AT EPOCH": 5081, "BSTAR": 9.2874e-5, "MEAN MOTION DOT": 2.462e-5, "MEAN MOTION DDOT": 0

🗊 🔒 https://celestrak.com/NORAD/elements/gp.php?NAME=NUSAT-8&FORMAT=XML

This XML file does not appear to have any style information associated with it. The document tree is shown below.

-<ndm xsi:noNamespaceSchemaLocation="https://sanaregistry.org/r/ndmxml/ndmxml-1.0-master.xsd"> -<omm id="CCSDS OMM VERS" version="2.0">

- -<header>
  - <CREATION\_DATE/>
  - <ORIGINATOR/>
- </header>

-<body>

← → C û

-<segment>

-<metadata>

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</metadata>

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<ARG\_OF\_PERICENTER>162.6718</ARG\_OF\_PERICENTER>

<MEAN\_ANOMALY>281.7748</MEAN\_ANOMALY>

</meanElements>

}]

# **4. Some details to consider** "Not everything that shines is made out of gold"...

5. OMM support in the open source ecosystem Could be better, but could also be worse

Name	OMM read	OMM write	Notes
python-sgp4	XML, CSV	In progress	Only elements, no covariances or metadata. JSON write in progress.
Orekit	KVN	Missing	
beyond	KVN, XML	KVN, XML	
python- <u>satellitetle</u>	In progress	In progress	JSON and CSV read in progress. XML and JSON write in progress.
oacmpy	KVN, XML	Missing	Only export to CZML.
ccsds-ndm	XML	XML	

## Conclusions

- It is already time to switch away from TLE/3LE!
- There is at least one open source library fully supporting OMMs
- Two-way conversion between the old and new format seems to be missing
- XML is hard, also for pros

## Thanks a lot!

<hello@juanlu.space>